

2.2.5 Discussion:

perihelion rotation predicted by GR for mercury
per century: $\Delta\phi = 43''$

observed perihelion rotation: $\Delta\phi = 574.6''$

subtract effect of other planets: $\Delta\phi = 531.5''$

discrepancy since Newcomb (1882): $43.1''$

← explained by GR

consequence: $\frac{2 - \beta + 2\gamma}{3} = 1.000 \pm 0.001$

• γ determined from light deflection: $\beta = 1.000 \pm 0.003$

• β stems from $(a/r)^2$ -term of Robertson expansion:
nonlinear effect of gravities

Perihelion shifts of other planets/asteroid:

(per century)	mercury	venus	earth	(cassini (0.5 ferm radius) 10 = 3'')
GR	43.03''	8.6''	3.8''	
observation	43.1'' \pm 0.5''	8.4'' \pm 3''	5.0'' \pm 2''	9.8 \pm 0.8''

$\epsilon \approx 1$ nearly circular

→ experimental uncertainty larger

$\epsilon \approx 0.8$

Physical origin of perihelion rotation:

additionally to Newton potential \approx GR predicts $\frac{1}{r^3}$ potential

2.2.6 Quadrupole Moment of Sun:

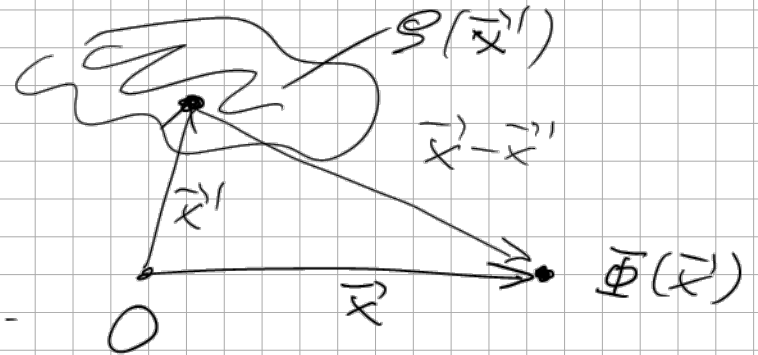
Newton gravity potential of a mass distribution

$$\Phi(\vec{x}) = -G \int d^3x' \frac{\rho(\vec{x}')}{|\vec{x} - \vec{x}'|}$$

mass density

approximation: $|\vec{x}| \gg |\vec{x}'|$

$$\frac{1}{|\vec{x} - \vec{x}'|} = \frac{1}{r} + \frac{x'_j}{r^3} x_j + \frac{1}{2} \frac{3x'_j x'_k - r^2 \delta_{jk}}{r^5} x'_j x'_k + \dots$$



$$\Phi(\vec{x}) = -G \left\{ \underbrace{\frac{M}{r}}_{\text{monopole}} + \underbrace{\frac{x'_j}{r^3} \int d^3x' \rho(\vec{x}') x'_j}_{\text{dipole}} + \underbrace{\frac{1}{2} \frac{3x'_j x'_k - r^2 \delta_{jk}}{r^5} T_{jk}}_{\text{quadrupole}} + \dots \right.$$

$$M = \int d^3x' \rho(\vec{x}')$$

assume $\rho(\vec{x}') = \rho(\vec{x}'')$

$$= 0$$

due to symmetry

quadrupole tensor

$$T_{jk} = \int d^3x' \rho(\vec{x}') x'_j x'_k$$

consider ellipsoid: ellipsoidal coordinates

$$x' = a \nu \sin \vartheta \cos \varphi \quad \nu \in (0, 1)$$

$$y' = b \nu \sin \vartheta \sin \varphi \quad \vartheta \in [0, \pi]$$

$$z' = c \nu \cos \vartheta \quad \varphi \in [0, 2\pi]$$

assuming $\frac{Q}{M}$ being constant in ellipsoid:
 $\Rightarrow M = \frac{4\pi}{3} \rho abc$

$$(T_{jk}) = \frac{1}{5} \frac{4\pi}{3} \rho abc \begin{pmatrix} a^2 & 0 & 0 \\ 0 & b^2 & 0 \\ 0 & 0 & c^2 \end{pmatrix} = \frac{M}{5} \begin{pmatrix} a^2 & & \\ & b^2 & \\ & & c^2 \end{pmatrix}$$

rotational ellipsoid: $a = b \neq c \Rightarrow$ effective radius $R = (a^2 c)^{1/3}$

$$\Phi(\vec{x}) = -\frac{GM}{r} - \frac{GM}{10} \frac{x^2 + y^2 - 2z^2}{(x^2 + y^2 + z^2)^{5/2}} (a^2 - c^2) + \dots$$

equatorial plane: $z = 0, r = \sqrt{x^2 + y^2}$

$$\Phi(\vec{x}) = -\frac{GM}{r} - \frac{GQ}{2r^3} + \dots, \quad Q = \frac{M}{5} \frac{a-c}{R} \cdot \frac{a+c}{R} R^2 \quad \text{quadrupole moment}$$

$\underbrace{\hspace{10em}}_{a \approx c \approx R}$

$\Rightarrow Q = q M R^2$, dimensionless asymmetry parameter

$$q = \frac{2}{5} \frac{a-c}{R}$$

Identify previous GR calculation with quadrupole moment

$$\underbrace{\frac{GM \ell^2}{m^2 c^2}} \sim \underbrace{GQ} \quad \text{with } \frac{\ell}{m} \sim \underbrace{v \rho}_{\text{parameter of ellipsoid}}, \quad v^2 = \frac{GM}{\rho}$$

prefactor of $1/r^3$ in GR \sim prefactor $1/r^3$ for quadrupole moment

$$\frac{G Q}{G M e^2 / m^2 c^2} \sim \frac{q R^2}{P a} \Rightarrow \Delta \phi = 3 \pi q \frac{R^2}{P^2}$$

helioseismology:

- sun is considered as a droplet, confined due to gravity
- rotation of sun = deformation of sphere to ellipsoid
- eigenfrequencies allow to deduce q
- 1960: Robert Leighton $\Rightarrow q = (1.7 \pm 0.4) \cdot 10^{-7}$

$$\Rightarrow \Delta \phi = 5 \cdot 10^{-5}''$$

This is $5 \cdot 10^{-4}$ smaller than perihelion of mercury due to GR

Outlook: perihelion shift of stars encircling the center of Milkyway since 2018

- mass of black hole $\approx 4 \cdot 10^6 M_{\odot}$
- one is trying to deduce spin (angular momentum) of black hole

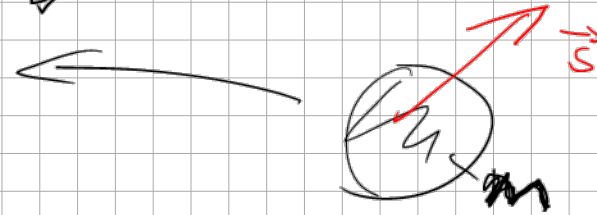
2.3 Geodetic Precession:

what is happening to a spinning top or gyroscope (high-precision

spinning top) is a static, isotropic gravitational field?

→ spin gets a rotation due to gravity

2.3.1 Equations of Motion



geodesic
trajectories

Schwarzschild
metric

spin
motion

$$\frac{d^2 x^\lambda}{d\tau^2} = - \Gamma_{\mu\nu}^\lambda \frac{dx^\mu}{d\tau} \frac{dx^\nu}{d\tau}$$

$$\frac{ds^\lambda}{d\tau} = - \Gamma_{\mu\nu}^\lambda s^\mu \frac{dx^\nu}{d\tau}$$

$$g_{\mu\nu} \frac{dx^\mu}{d\tau} \frac{dx^\nu}{d\tau} = c^2 \quad (**)$$

$$g_{\mu\nu} s^\mu s^\nu = -s^2 \quad \Leftarrow \text{normalization}$$

$$g_{\mu\nu} \frac{dx^\mu}{d\tau} s^\nu = 0 \quad \text{orthogonality}$$

$$\frac{d^2 x^0}{d\tau^2} = - \frac{B'(r)}{B(r)} \frac{dx^0}{d\tau} \frac{dr}{d\tau} \quad \checkmark$$

$$\frac{d^2 r}{d\tau^2} = - \frac{B'(r)}{2A(r)} \left(\frac{dx^0}{d\tau}\right)^2 - \frac{A'(r)}{2A(r)} \left(\frac{dr}{d\tau}\right)^2 + \frac{r}{A(r)} \left(\frac{d\varphi}{d\tau}\right)^2 + \frac{r}{A(r)} \sin^2 \vartheta \left(\frac{d\varphi}{d\tau}\right)^2 \quad (*)$$

$$\frac{d^2 \vartheta}{d\tau^2} = - \frac{r}{2} \frac{dr}{d\tau} \frac{d\vartheta}{d\tau} + r \sin \vartheta \cos \vartheta \left(\frac{d\varphi}{d\tau}\right)^2 \quad \checkmark$$

$$\frac{d^2 \varphi}{d\tau^2} = - \frac{r}{2} \frac{dr}{d\tau} \frac{d\varphi}{d\tau} - 2 \frac{\cos \vartheta}{\sin^3 \vartheta} \frac{dr}{d\tau} \frac{d\vartheta}{d\tau} \quad \checkmark$$

$$\frac{ds^0}{d\tau} = -\frac{B'(r)}{2B(r)} \left(\cancel{s^0 \frac{d\tau}{d\tau}} + s^1 \frac{dx^0}{d\tau} \right)$$

$$\frac{ds^1}{d\tau} = -\frac{B'(r)}{2A(r)} s^0 \frac{dx^0}{d\tau} - \frac{A'(r)}{2A(r)} r \frac{dr}{d\tau} + \frac{2}{r} s^2 \frac{d\varphi}{d\tau} + \frac{2}{r} \sin^2 \vartheta s^3 \frac{d\varphi}{d\tau}$$

$$\frac{ds^2}{d\tau} = -\frac{1}{2} \left(\cancel{s^1 \frac{d\vartheta}{d\tau}} + \cancel{s^2 \frac{d\vartheta}{d\tau}} \right) + \cancel{\sin \vartheta \cos \vartheta} s^3 \frac{d\varphi}{d\tau}$$

$$\frac{ds^3}{d\tau} = -\frac{1}{2} \left(s^1 \frac{d\varphi}{d\tau} + \cancel{s^2 \frac{d\varphi}{d\tau}} \right) - \frac{\cos \vartheta}{\sin \vartheta} \left(\cancel{s^2 \frac{d\varphi}{d\tau}} + \cancel{s^3 \frac{d\varphi}{d\tau}} \right)$$

2.3.2 Satellite Trajectory in Form of a Circle:

$$(x^\mu(\tau), r(\tau), \vartheta(\tau), \varphi(\tau)) = (u^0 \tau, \underbrace{r}_{\text{fixed}}, \frac{\pi}{2}, \omega_0 \tau)$$

$$\left(\frac{dx^0}{d\tau}, \frac{dr}{d\tau}, \frac{d\vartheta}{d\tau}, \frac{d\varphi}{d\tau} \right) = (u^0, \underline{0}, \underline{0}, \omega_0)$$

(*) provides a relation between three parameters u^0, r, ω_0

in our ansatz for a circular motion:

$$(*) \Rightarrow \left(\frac{u^0}{\omega_0} \right)^2 = \frac{r^2}{B(r)} \quad (1)$$

(**) leads to another condition for 3 parameters:

$$B(r)(u^0)^2 - r^2 \omega_0^2 = c^2 \quad (2)$$

(1) + (2) in order to eliminate u^0 :

$$\omega_0^2 = \frac{c^2}{z \sqrt{\frac{B(z)}{B'(z)} - z^2}} = \frac{ac^2}{z^3 - 3az^2}, \quad a = \frac{GM}{c^2} = \frac{25}{2}$$

Schwarzschild: $B(z) = 1 - \frac{2a}{z}$

earth radius: $R_E = 6.4000 \text{ km}$
 earth acceleration: $g = \frac{GM}{R_E^2}$ } $a = 4.6 \text{ mm}$

$$\omega_0^2 \approx \frac{ac^2}{R_E^3} = \frac{GM}{R_E^3}$$

known from mechanics:

$$\frac{mGM}{R_E^2} = m\omega_0^2 R_E$$

$$T_0 = \frac{2\pi}{\omega_0} = 2\pi \sqrt{\frac{R_E}{g}} \approx 1.4 \text{ h}$$

spin equation of motion for a spin in satellite:

$$\frac{ds^0}{d\tau} = - \frac{B'(z)}{2B(z)} u^0 s^1$$

$$\frac{ds^1}{d\tau} = - \frac{B'(z)}{2B(z)} u^0 s^0 + \frac{z}{r(z)} \omega_0 s^3$$

$$\frac{ds^2}{d\tau} = 0 \Rightarrow s^2(\tau) = \text{const.} = 0$$

four linear ordinary differential equations with constant coefficients

$$\frac{dS^3}{d\tau} = -\frac{1}{2} \omega_0 S^2$$

initial condition: $(S^0(0), S^1(0), S^2(0), S^3(0)) = (0, s, 0, 0)$

$$g_{mn} S^m_{(0)} u^{\nu}_{(0)} = 0$$

2.3.3 Solution for Spin Dynamics:

$\Rightarrow S^2(\tau) = 0 \Rightarrow$ spatial spin vector must lie within equatorial plane

$$\vec{r} = r \begin{pmatrix} \sin\delta \cos\phi \\ \sin\delta \sin\phi \\ \cos\delta \end{pmatrix}, \quad \vec{e}_\delta = \frac{\partial \vec{r}}{\partial \delta} = \frac{\partial \vec{r}}{\partial \delta} \Big|_{\delta=\pi/2} = \begin{pmatrix} 0 \\ 0 \\ -1 \end{pmatrix}$$

Combining remaining 3 eqs.

$$\frac{d^2 S^1}{d\tau^2} = -\omega^2 S^1, \quad \omega = \omega_0 \sqrt{\frac{1}{A(\omega)} \left[1 - \frac{2B(\omega)}{2A(\omega)} \right]} \\ \approx \omega_0 \left\{ 1 - \left(\gamma + \frac{1}{2} \right) \frac{a}{r} + \dots \right\}$$

ω_0 : angular frequency for center-of-mass motion

ω : " " of spin rotation